

SCE 594: Special Topics in Intelligent Automation & Robotics

Lecture 23: Stabilization Control on $SO(3)$ III and $SE(3)$



Outline

- Recap last lectures
- Lyapunov analysis of closed loop system
- From $SO(3)$ to $SE(3)$
- From $SE(3)$ to MAV control



Outline

- Recap last lectures
- Lyapunov analysis of closed loop system
- From $SO(3)$ to $SE(3)$
- From $SE(3)$ to MAV control



Recap: Rigid Body Rotation Dynamics

- The equations of a rotating rigid body with control torques τ are:

- $x = (R, p) \in SO(3) \times \mathbb{R}^3$

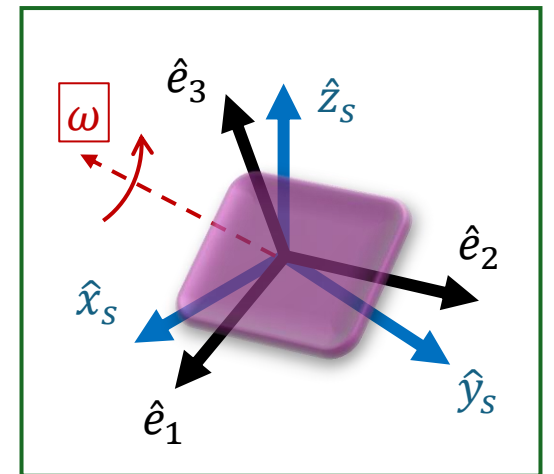
- $\begin{pmatrix} \dot{R} \\ \dot{p} \end{pmatrix} = \begin{pmatrix} \beta_R(\omega) \\ \tilde{p} \omega \end{pmatrix} + \begin{pmatrix} 0 \\ I_3 \end{pmatrix} \tau$

with $\omega = J^{-1}p$

and $\beta_R: \mathbb{R}^3 \rightarrow T_R SO(3), \quad \omega \mapsto \beta_R(\omega) := R \tilde{\omega}$

- Total energy of the system

- $H(x) = H(p) = \frac{1}{2} p^\top J^{-1} p$



Recap: Stabilization Control on $SO(3)$

- We aim to design a **Geometric** PD controller

$$\tau = \tau_p + \tau_d$$

such that $x_d = (R_d, 0)$ is an asymptotically stable equilibrium point of the closed loop system

$$\begin{pmatrix} \dot{R} \\ \dot{p} \end{pmatrix} = \begin{pmatrix} \beta_R(\omega) \\ \tilde{p} \omega + \tau_p + \tau_d \end{pmatrix}.$$

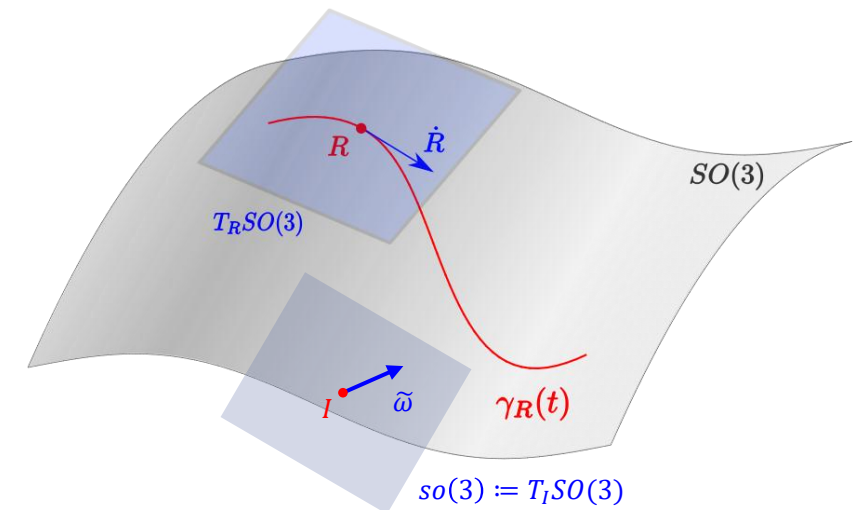
gradient of some pos.def. function $\Psi(R)$
on $SO(3)$ with a minimum at $R = R_d$.

Inject damping such that time derivative
of $V(x) = H(p) + \Psi(R)$ is neg. def.



Recap: Geometric structure of $SO(3)$

- The geometric nature of $SO(3)$ will be reflected in
 1. How to compute the error between $R, R_d \in SO(3)$?
 2. How to design $\Psi(R)$ to be positive definite ?
 3. How to compute $d\Psi(R) \in T_R^*SO(3)$?
 4. How to convert $d\Psi(R)$ to the proportional torque $\tau_p \in \mathbb{R}^3$?
 5. How to design the derivative torque $\tau_d \in \mathbb{R}^3$?



Recap: Stabilization Control on $SO(3)$

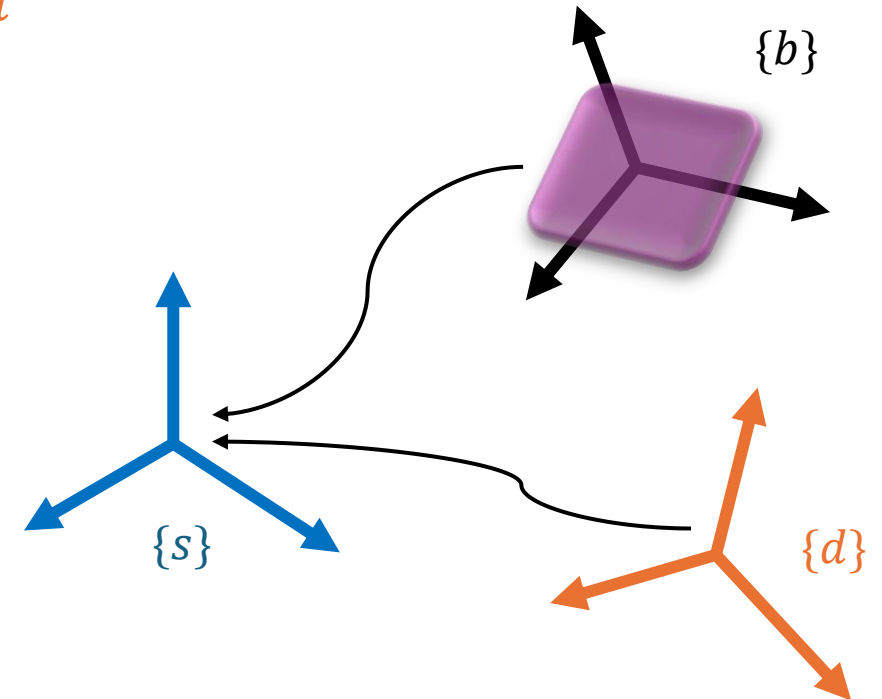
1. How to compute the error between $R, R_d \in SO(3)$?

- Error between actual and desired

$$R_e := R_d^T R \in SO(3)$$

- We have that

$$R_e \rightarrow I \text{ as } R \rightarrow R_d$$



Recap: Stabilization Control on $SO(3)$

2. How to design $\Psi(R)$ to be positive definite ?

- One choice for $\Psi: SO(3) \rightarrow \mathbb{R}$ is

$$\Psi(R) := \frac{1}{2} \text{tr}(I - R_d^T R)$$

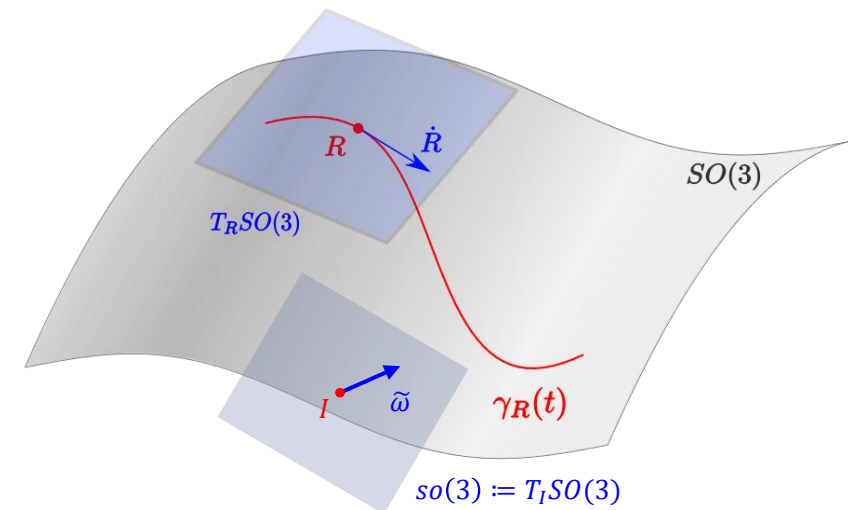
- $\Psi(R_d) = 0$
- We have shown before that it can be written as

$$\Psi(R) = 1 - \cos \theta$$

for some $\theta \in (-\pi, \pi]$.

- Therefore,

$$0 \leq \Psi(R) \leq 2$$



Recap: Stabilization Control on $SO(3)$

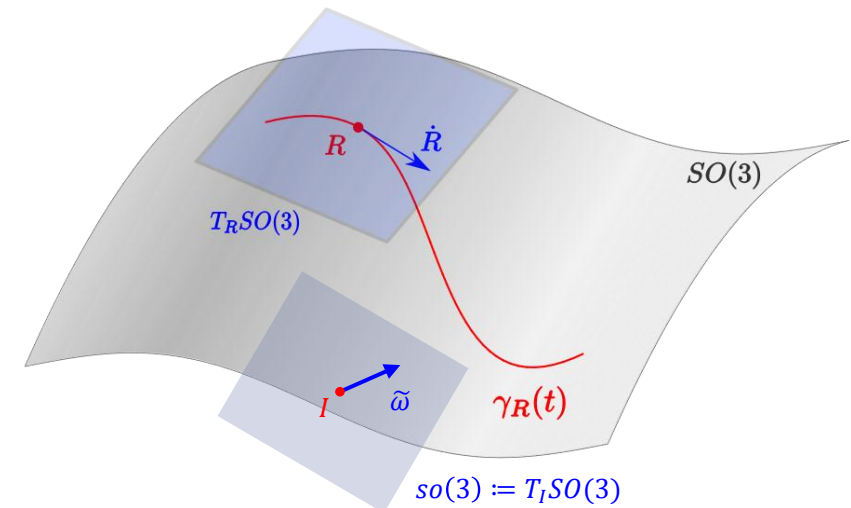
3. How to compute $d\Psi(R) \in T_R^*SO(3)$?

- The differential of $\Psi: SO(3) \rightarrow \mathbb{R}$ is defined as the unique covector $d\Psi(R) \in T_R^*SO(3)$ that satisfies

$$\langle d\Psi(R) | \delta R \rangle_{T_R SO(3)} = \left. \frac{d}{d\epsilon} \right|_{\epsilon=0} \Psi(R_\epsilon), \quad \forall \delta R \in T_R SO(3)$$

where R_ϵ is a curve on $SO(3)$ that satisfies

$$R_\epsilon|_{\epsilon=0} = R, \quad \left. \frac{d}{d\epsilon} \right|_{\epsilon=0} R_\epsilon = \delta R$$



The pairing between any covector in $T_R^*SO(3)$ and vector $T_R SO(3)$ is:
 $\langle \Gamma | \delta R \rangle_{T_R SO(3)} := \text{tr}(\Gamma \delta R)$



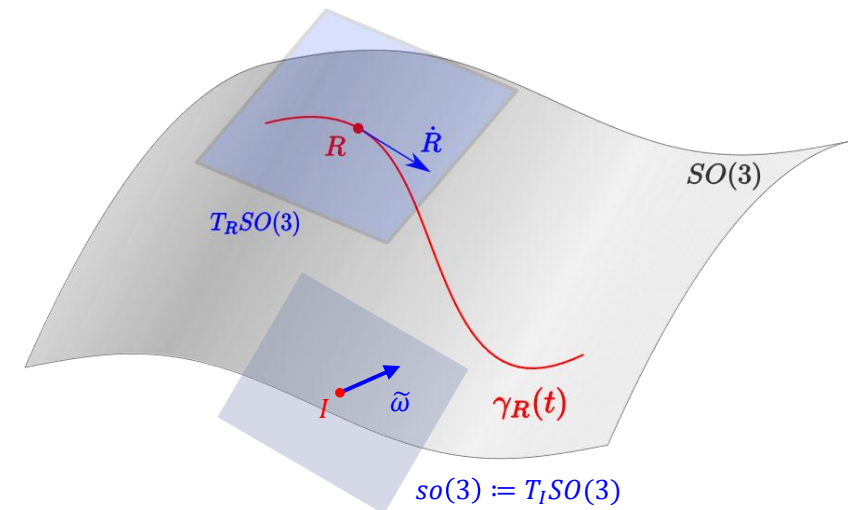
Recap: Stabilization Control on $SO(3)$

3. How to compute $d\Psi(R) \in T_R^*SO(3)$?

- $\Psi(\mathbf{R}_\epsilon) = \frac{1}{2} \text{tr} (I - R_d^\top \mathbf{R}_\epsilon)$
- $\frac{d}{d\epsilon} \Big|_{\epsilon=0} \Psi(\mathbf{R}_\epsilon) = -\frac{1}{2} \text{tr} \left(R_d^\top \frac{d}{d\epsilon} \Big|_{\epsilon=0} \mathbf{R}_\epsilon \right) = -\frac{1}{2} \text{tr} (R_d^\top \delta R)$
- $\langle d\Psi(R) | \delta R \rangle_{T_R SO(3)} = \text{tr} (d\Psi(R) \delta R)$

- By comparing both sides, we have that

$$d\Psi(R) = -\frac{1}{2} R_d^\top$$

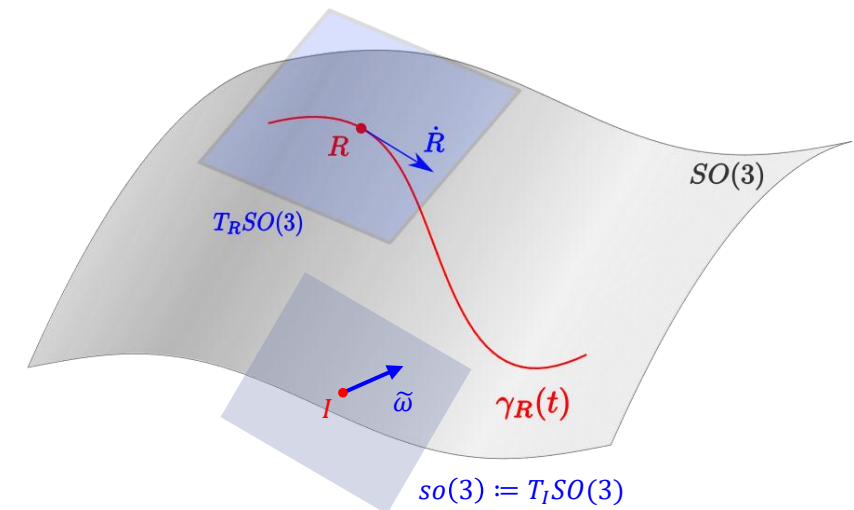


Recap: Stabilization Control on $SO(3)$

4. How to convert $d\Psi(R)$ to the proportional torque $\tau_p \in \mathbb{R}^3$?

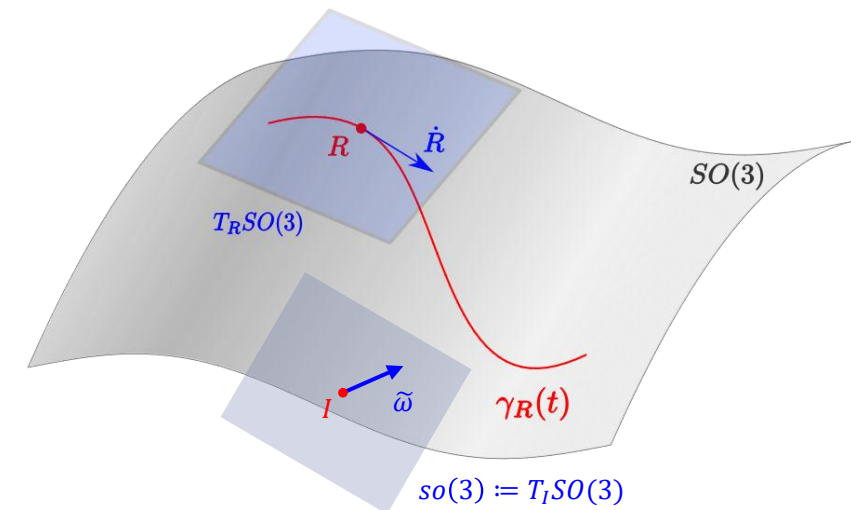
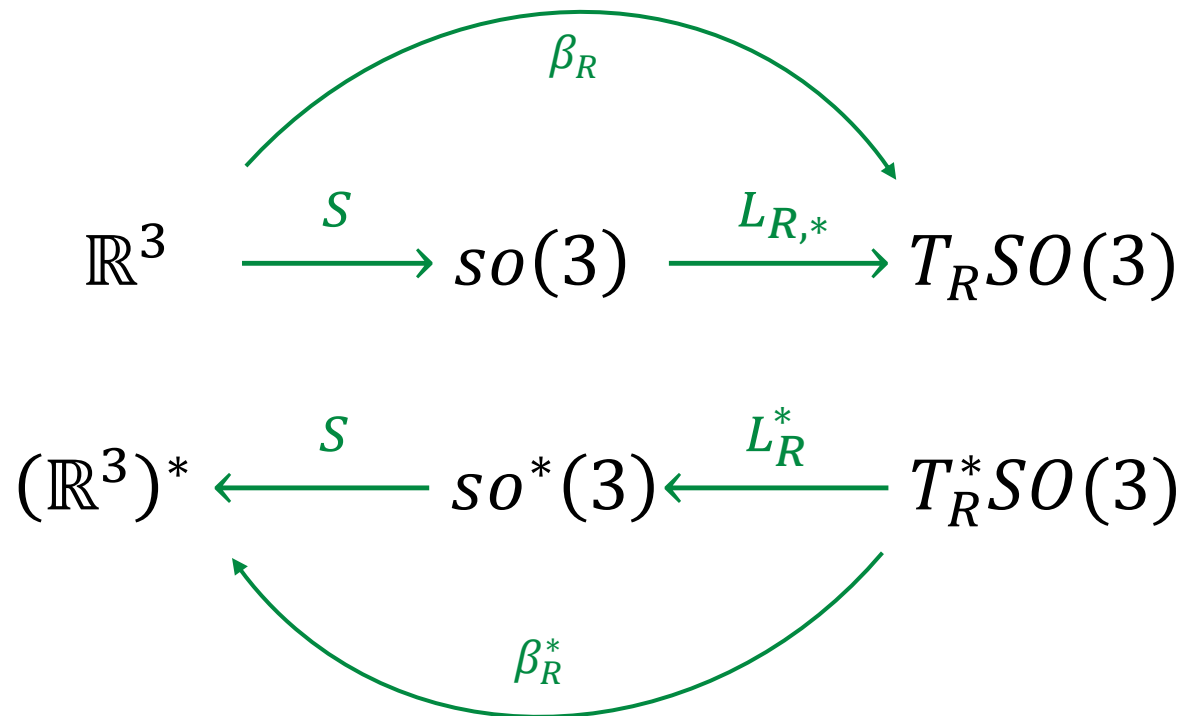
- We use the duality of linear maps !
- Let $A: U \rightarrow V$ be a linear map between the vector spaces U and V . Then its dual map $A^*: V^* \rightarrow U^*$ is defined (implicitly) by:

$$\langle A^*(\alpha) | u \rangle_U = \langle \alpha | A(u) \rangle_V, \quad \forall u \in U, \alpha \in V^*$$



Recap: Stabilization Control on $SO(3)$

4. How to convert $d\Psi(R)$ to the proportional torque $\tau_p \in \mathbb{R}^3$?

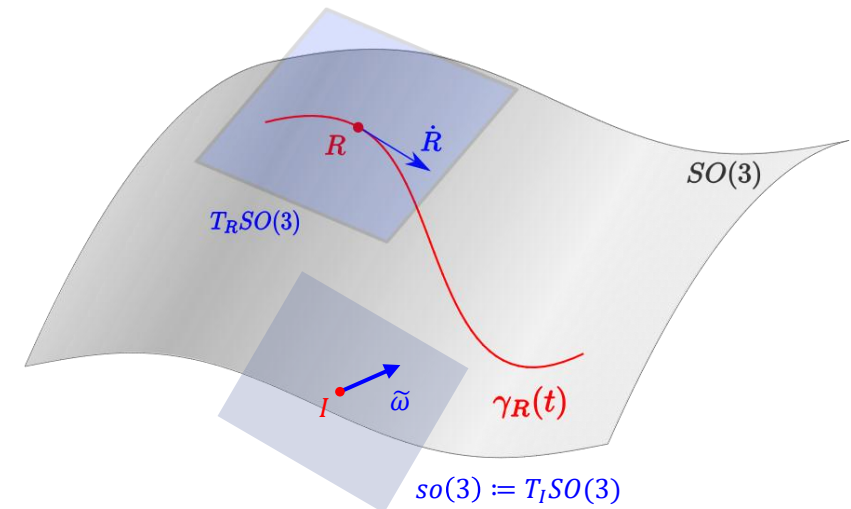


Recap: Stabilization Control on $SO(3)$

4. How to convert $d\Psi(R)$ to the proportional torque $\tau_p \in \mathbb{R}^3$?

- $d\Psi(R) = -\frac{1}{2}R_d^\top \in T_R^*SO(3)$
- $\tilde{e}_R := L_R^*(d\Psi(R)) = \text{sk}(R_d^\top R) = \frac{1}{2}(R_d^\top R - R^\top R_d) \in \mathfrak{so}^*(3)$
- $e_R := \beta_R^*(d\Psi(R)) = S^{-1}(\tilde{e}_R) = \frac{1}{2}S^{-1}(R_d^\top R - R^\top R_d) \in (\mathbb{R}^3)^*$
- The proportional torque is then designed as

$$\tau_p := -k_p e_R$$



Proportional gain: $k_p \in \mathbb{R}_+$



Recap: Stabilization Control on $SO(3)$

5. How to design the derivative torque $\tau_d \in \mathbb{R}^3$?

- Let's conduct a Lyapunov analysis for the closed loop system

$$\begin{pmatrix} \dot{R} \\ \dot{p} \end{pmatrix} = \begin{pmatrix} \beta_R(\omega) \\ \tilde{p} \omega + \tau_p + \tau_d \end{pmatrix} = \begin{pmatrix} \beta_R(\omega) \\ \tilde{p} \omega - k_p \beta_R^*(d\Psi(R)) + \tau_d \end{pmatrix}$$



Outline

- Recap last lectures
- Lyapunov analysis of closed loop system
- From $SO(3)$ to $SE(3)$
- From $SE(3)$ to MAV control



Stabilization Control on $SO(3)$

- Consider the Lyapunov function:

$$V(R, p) = H(p) + k_p \Psi(R)$$

where

$$H(p) := \frac{1}{2} p^\top J^{-1} p, \quad \text{and} \quad \Psi(R) := \frac{1}{2} \text{tr}(I - R_d^\top R)$$

with

$$dH(p) = (J^{-1} p)^\top = \omega^\top, \quad \text{and} \quad d\Psi(R) = -\frac{1}{2} R_d^\top$$



Stability Analysis

Compute the time derivative of V

- $V(R, p) = H(p) + k_p \Psi(R)$

- $\dot{V}(R, p) = \dot{H}(p) + k_p \dot{\Psi}(R) = \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + k_p \langle d\Psi(R) | \dot{R} \rangle_{T_R SO(3)}$



Stability Analysis

Compute the time derivative of V

- $V(R, p) = H(p) + k_p \Psi(R)$

- $\dot{V}(R, p) = \dot{H}(p) + k_p \dot{\Psi}(R) = \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + k_p \langle d\Psi(R) | \dot{R} \rangle_{T_R SO(3)}$
 $= \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + k_p \langle d\Psi(R) | \beta_R(\omega) \rangle_{T_R SO(3)}$



Stability Analysis

Compute the time derivative of V

- $V(R, p) = H(p) + k_p \Psi(R)$
- $\dot{V}(R, p) = \dot{H}(p) + k_p \dot{\Psi}(R) = \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + k_p \langle d\Psi(R) | \dot{R} \rangle_{T_{R}SO(3)}$
$$= \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + k_p \langle d\Psi(R) | \beta_R(\omega) \rangle_{T_{R}SO(3)}$$
$$= \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + k_p \langle \beta_R^*(d\Psi(R)) | \omega \rangle_{\mathbb{R}^3}$$
$$= \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + \langle k_p e_R | \omega \rangle_{\mathbb{R}^3}$$

$$e_R := \beta_R^*(d\Psi(R))$$

Dual map: $\langle A^*(\alpha) | u \rangle_U = \langle \alpha | A(u) \rangle_V$



Stability Analysis

Compute the time derivative of V

- $V(R, p) = H(p) + k_p \Psi(R)$

- $\dot{V}(R, p) = \dot{H}(p) + k_p \dot{\Psi}(R) = \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + \langle k_p e_R | \omega \rangle_{\mathbb{R}^3}$



Stability Analysis

Compute the time derivative of V

- $V(R, p) = H(p) + k_p \Psi(R)$

- $\dot{V}(R, p) = \dot{H}(p) + k_p \dot{\Psi}(R) = \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + \langle k_p e_R | \omega \rangle_{\mathbb{R}^3}$
 $= \omega^\top (\tilde{p} \omega - k_p e_R + \tau_d) + \omega^\top k_p e_R$
 $= \omega^\top \tilde{p} \omega - \omega^\top (k_p - k_p) e_R + \omega^\top \tau_d$

$$e_R := \beta_R^*(d\Psi(R))$$

$$\tau_p = -K_p e_R$$



Stability Analysis

Compute the time derivative of V

- $V(R, p) = H(p) + k_p \Psi(R)$

- $\dot{V}(R, p) = \dot{H}(p) + k_p \dot{\Psi}(R) = \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + \langle k_p e_R | \omega \rangle_{\mathbb{R}^3}$
$$= \omega^\top (\tilde{p} \omega - k_p e_R + \tau_d) + \omega^\top k_p e_R$$
$$= \omega^\top \tilde{p} \omega - \omega^\top (k_p - k_p) e_R + \omega^\top \tau_d$$
$$= \underbrace{\omega^\top \tilde{p} \omega}_{=0} - \underbrace{\omega^\top (k_p - k_p) e_R}_{=0} + \omega^\top \tau_d$$
$$= \omega^\top \tau_d$$

$$e_R := \beta_R^*(d\Psi(R))$$

$$\tau_p = -K_p e_R$$



Stability Analysis

Compute the time derivative of V

- $V(R, p) = H(p) + k_p \Psi(R)$

- $\dot{V}(R, p) = \dot{H}(p) + k_p \dot{\Psi}(R) = \langle dH(p) | \dot{p} \rangle_{\mathbb{R}^3} + \langle k_p e_R | \omega \rangle_{\mathbb{R}^3}$

$$= \omega^\top (\tilde{p} \omega - k_p e_R + \tau_d) + \omega^\top k_p e_R$$

$$= \omega^\top \tilde{p} \omega - \omega^\top (k_p - k_p) e_R + \omega^\top \tau_d$$

$$= \underbrace{\omega^\top \tilde{p} \omega}_{=0} - \underbrace{\omega^\top (k_p - k_p) e_R}_{=0} + \omega^\top \tau_d$$

$$= \omega^\top \tau_d = -\omega^\top K_d \omega \leq 0$$

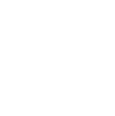
neg. semi-def.

Choose $\tau_d = -K_d \omega$ with $K_d > 0$



La Salle's Invariance Principle

- Let $\Omega = \mathcal{X} = SO(3) \times \mathbb{R}^3$
- $\mathcal{R} := \{(R, p) \in \Omega \mid \dot{V}(R, p)\}$
 - $\dot{V}(R, p) = \omega^\top K_d \omega \quad \Rightarrow \omega = p = 0$



La Salle's Invariance Principle

- Let $\Omega = \mathcal{X} = SO(3) \times \mathbb{R}^3$
- $\mathcal{R} := \{(R, p) \in \Omega \mid \dot{V}(R, p)\}$
 - $\dot{V}(R, p) = \omega^\top K_d \omega \quad \Rightarrow \omega = p = 0$
- $\mathcal{M} := \{(R, p) \in \mathcal{R} \mid \dot{p} = 0\}$
 - $\dot{p} = \tilde{p} \omega - k_p e_R - K_d \omega$



La Salle's Invariance Principle

- Let $\Omega = \mathcal{X} = SO(3) \times \mathbb{R}^3$
- $\mathcal{R} := \{(R, p) \in \Omega \mid \dot{V}(R, p)\}$
 - $\dot{V}(R, p) = \omega^\top K_d \omega \implies \omega = p = 0$
- $\mathcal{M} := \{(R, p) \in \mathcal{R} \mid \dot{p} = 0\}$
 - $\dot{p} = \tilde{p} \omega - k_p e_R - K_d \omega \implies e_R = 0$
 - $\implies \tilde{e}_R = \text{sk}(R_d^\top R) = 0$
 - $\implies R_e - R_e^\top = 0$
 - $\implies R_e = R_e^\top$



La Salle's Invariance Principle

- Let $\Omega = \mathcal{X} = SO(3) \times \mathbb{R}^3$
- $\mathcal{R} := \{(R, p) \in \Omega \mid \dot{V}(R, p)\}$
 - $\dot{V}(R, p) = \omega^\top K_d \omega \implies \omega = p = 0$
- $\mathcal{M} := \{(R, p) \in \mathcal{R} \mid \dot{p} = 0\}$
 - $\dot{p} = \tilde{p} \omega - k_p e_R - K_d \omega \implies e_R = 0$
 - $\implies \tilde{e}_R = \text{sk}(R_d^\top R) = 0$
 - $\implies R_e - R_e^\top = 0$
 - $\implies R_e = R_e^\top$

Therefore, largest invariant set in \mathcal{X} is:

$$\mathcal{M} := \{(R, p) \in \mathcal{R} \mid p = 0, R_d^\top R = R R_d^\top\}$$

How many $R \in SO(3)$ satisfy this ?



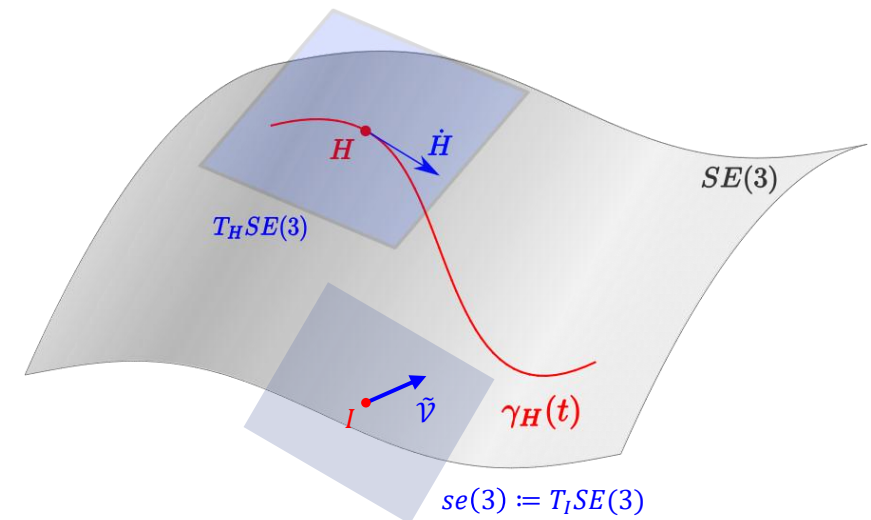
Outline

- Recap last lectures
- Lyapunov analysis of closed loop system
- **From $SO(3)$ to $SE(3)$**
- From $SE(3)$ to MAV control



Geometric structure of $SE(3)$

- The geometric nature of $SE(3)$ will be reflected in
 1. How to compute the error between $H, H_d \in SE(3)$?
 2. How to design $\Psi(H)$ to be positive definite ?
 3. How to compute $d\Psi(H) \in T_H^*SE(3)$?
 4. How to convert $d\Psi(H)$ to the proportional wrench $\mathcal{W}_p \in (\mathbb{R}^6)^*$?
 5. How to design the derivative wrench $\mathcal{W}_d \in (\mathbb{R}^6)^*$?



Stabilization Control on $SE(3)$

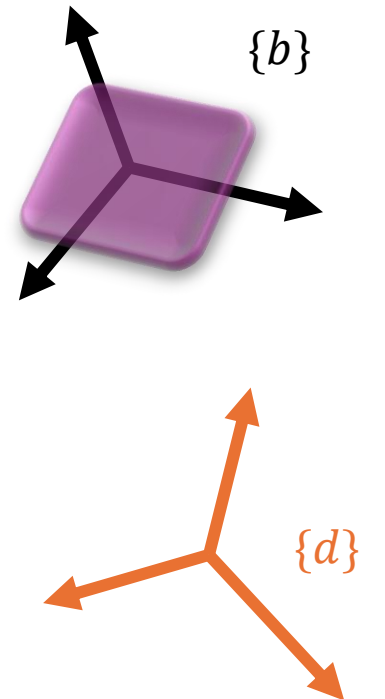
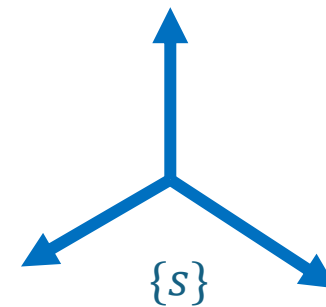
1. How to compute the error between $H, H_d \in SE(3)$?

- Error between actual and desired

$$H_e := H_d^{-1}H \in SE(3)$$

- We have that

$$H_e \rightarrow I \text{ as } H \rightarrow H_d$$



Stabilization Control on $SE(3)$

2. How to design $\Psi(H)$ to be positive definite ?

- One choice for $\Psi: SE(3) \rightarrow \mathbb{R}$ is to split rotation and translation

$$\Psi(H) = \Psi_t(\xi) + \Psi_o(R) = \frac{1}{2} (\xi - \xi_d)^\top K_t (\xi - \xi_d) + \frac{1}{2} \text{tr} \left(G_o (I - R_d^\top R) \right)$$



Stabilization Control on $SE(3)$

2. How to design $\Psi(H)$ to be positive definite ?

- One choice for $\Psi: SE(3) \rightarrow \mathbb{R}$ is to split rotation and translation

$$\Psi(H) = \Psi_t(\xi) + \Psi_o(R) = \frac{1}{2} (\xi - \xi_d)^\top K_t (\xi - \xi_d) + \frac{1}{2} \text{tr} \left(G_o (I - R_d^\top R) \right)$$

Table 1

Error functions and transport elements on $SE(3)$

Error function	Transport element	Comments
$\phi_1(R_d^\top R) + \frac{1}{2} \ p - p_d\ _{K_2}^2$	$(R^\top R_d, 0)$	$\phi_1(g_{e,1})$, not invariant, symmetric, gains expressed in inertial frame
$\phi_1(R_d^\top R) + \frac{1}{2} \ R_d^\top (p - p_d)\ _{K_2}^2$	$g^{-1} g_d$	$\phi_1(g_{e,r})$, invariant, not symmetric, gains expressed in reference frame
$\phi_1(R_d^\top R) + \frac{1}{2} \ R^\top (p - p_d)\ _{K_2}^2$	$g^{-1} g_d$	not symmetric, gains expressed in body frame
$\phi_1(R_d^\top R) + \frac{1}{2} \ (R^\top + R_d^\top)(p - p_d)\ _{K_2}^2$	$g^{-1} g_d$	$\phi_3(g_{e,r})$, invariant, symmetric
$\phi_1(RR_d^\top) + \frac{1}{2} \ (R + R_d)(p - p_d)\ _{K_2}^2$	$(I_3, 0)$	$\phi_3(g_{e,l})$, not invariant, symmetric
$\phi_1(RR_d^\top) + \frac{1}{2} \ R^\top p - R_d^\top p_d\ _{K_2}^2$	$(I_3, R_d^\top p_d - R^\top p)$	$\phi_1(g_{e,2})$, not invariant, symmetric



Stabilization Control on $SE(3)$

2. How to design $\Psi(H)$ to be positive definite ?

- One choice for $\Psi: SE(3) \rightarrow \mathbb{R}$ is to split rotation and translation

$$\Psi(H) = \Psi_t(\xi) + \Psi_o(R) = \frac{1}{2} (\xi - \xi_d)^\top K_t (\xi - \xi_d) + \frac{1}{2} \text{tr} \left(G_o (I - R_d^\top R) \right)$$

Table 1

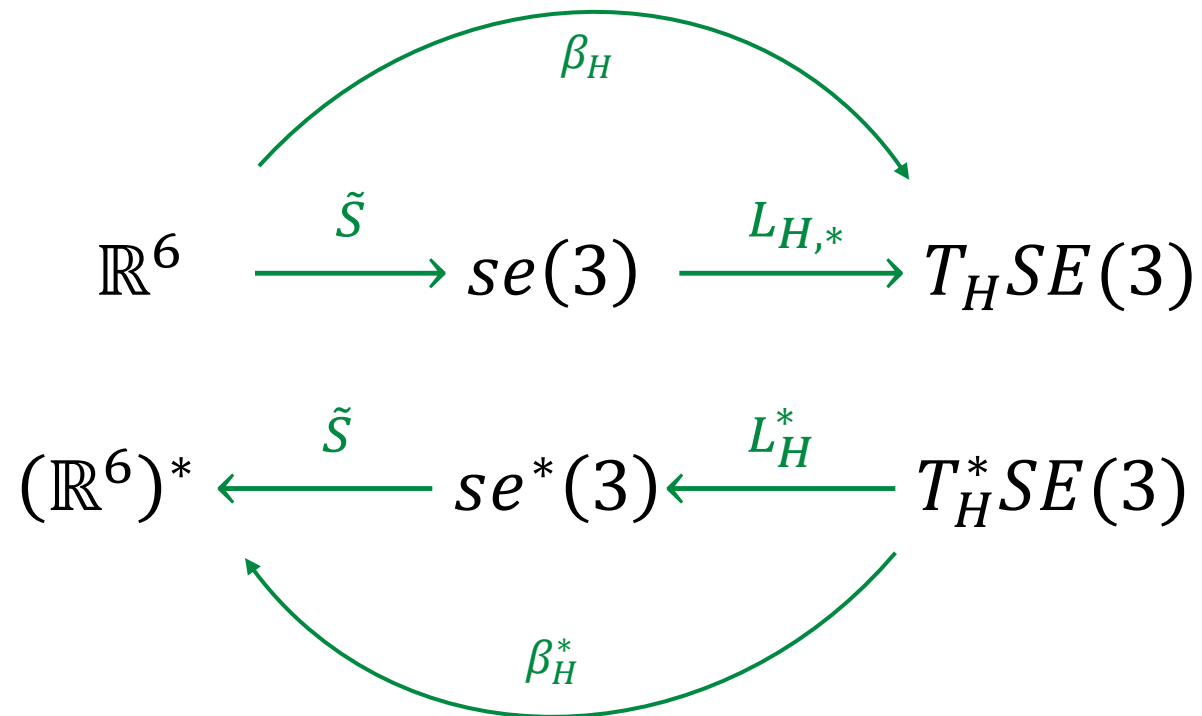
Error functions and transport elements on $SE(3)$

Error function	Transport element	Comments
$\phi_1(R_d^\top R) + \frac{1}{2} \ p - p_d\ _{K_2}^2$	$(R^\top R_d, 0)$	$\phi_1(g_{e,1})$, not invariant, symmetric, gains expressed in inertial frame
$\phi_1(R_d^\top R) + \frac{1}{2} \ R_d^\top (p - p_d)\ _{K_2}^2$	$g^{-1} g_d$	$\phi_1(g_{e,r})$, invariant, not symmetric, gains expressed in reference frame
$\phi_1(R_d^\top R) + \frac{1}{2} \ R^\top (p - p_d)\ _{K_2}^2$	$g^{-1} g_d$	not symmetric, gains expressed in body frame
$\phi_1(R_d^\top R) + \frac{1}{2} \ (R^\top + R_d^\top)(p - p_d)\ _{K_2}^2$	$g^{-1} g_d$	$\phi_3(g_{e,r})$, invariant, symmetric
$\phi_1(RR_d^\top) + \frac{1}{2} \ (R + R_d)(p - p_d)\ _{K_2}^2$	$(I_3, 0)$	$\phi_3(g_{e,l})$, not invariant, symmetric
$\phi_1(RR_d^\top) + \frac{1}{2} \ R^\top p - R_d^\top p_d\ _{K_2}^2$	$(I_3, R_d^\top p_d - R^\top p)$	$\phi_1(g_{e,2})$, not invariant, symmetric



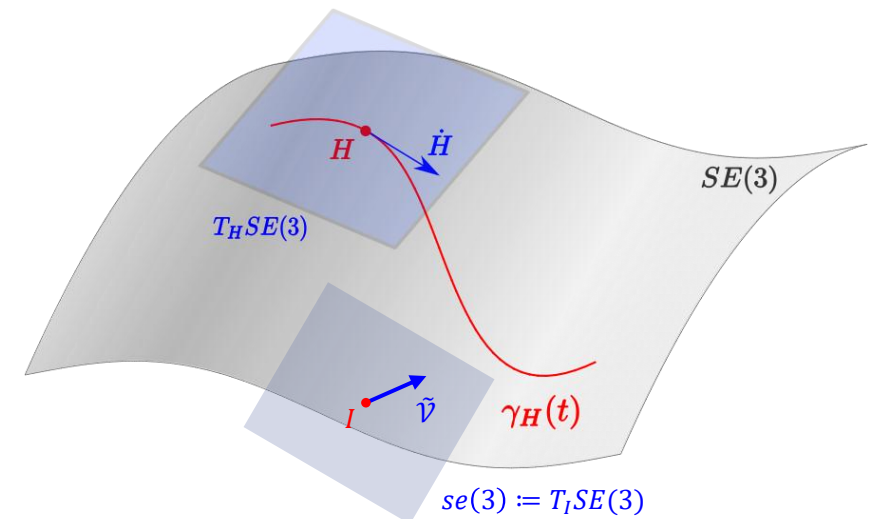
Stabilization Control on $SE(3)$

4. How to convert $d\Psi(H)$ to proportional wrench $\mathcal{W}_p \in (\mathbb{R}^6)^*$?



$$\mathcal{W}_p := \beta_H^*(d\Psi(H))$$

$$\beta_H(\delta\mathcal{V}) := H \delta\tilde{\mathcal{V}}$$



$$\langle \beta_H^*(d\Psi(H)) | \delta\mathcal{V} \rangle = \langle d\Psi(H) | \beta_H(\delta\mathcal{V}) \rangle$$



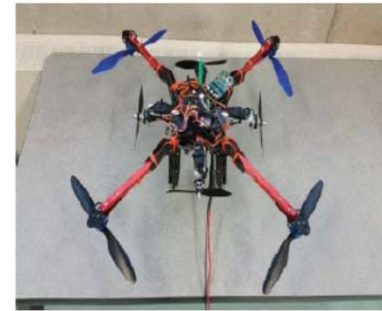
Outline

- Recap last lectures
- Lyapunov analysis of closed loop system
- From $SO(3)$ to $SE(3)$
- From $SE(3)$ to MAV control



Multi-rotor aerial vehicles

- Multi-rotor aerial vehicles (MAVs) are the most popular choice of aerial robotics platform.
- Usually, they have a simple mechanical structure and few moving parts.
- MAVs are usually modeled as a single rigid body floating in space.



Multi-rotor aerial vehicles

- MAVs are classified based on properties of the map between individual rotor thrusts λ_i and the resultant wrench applied on the MAV's body which is used for control.

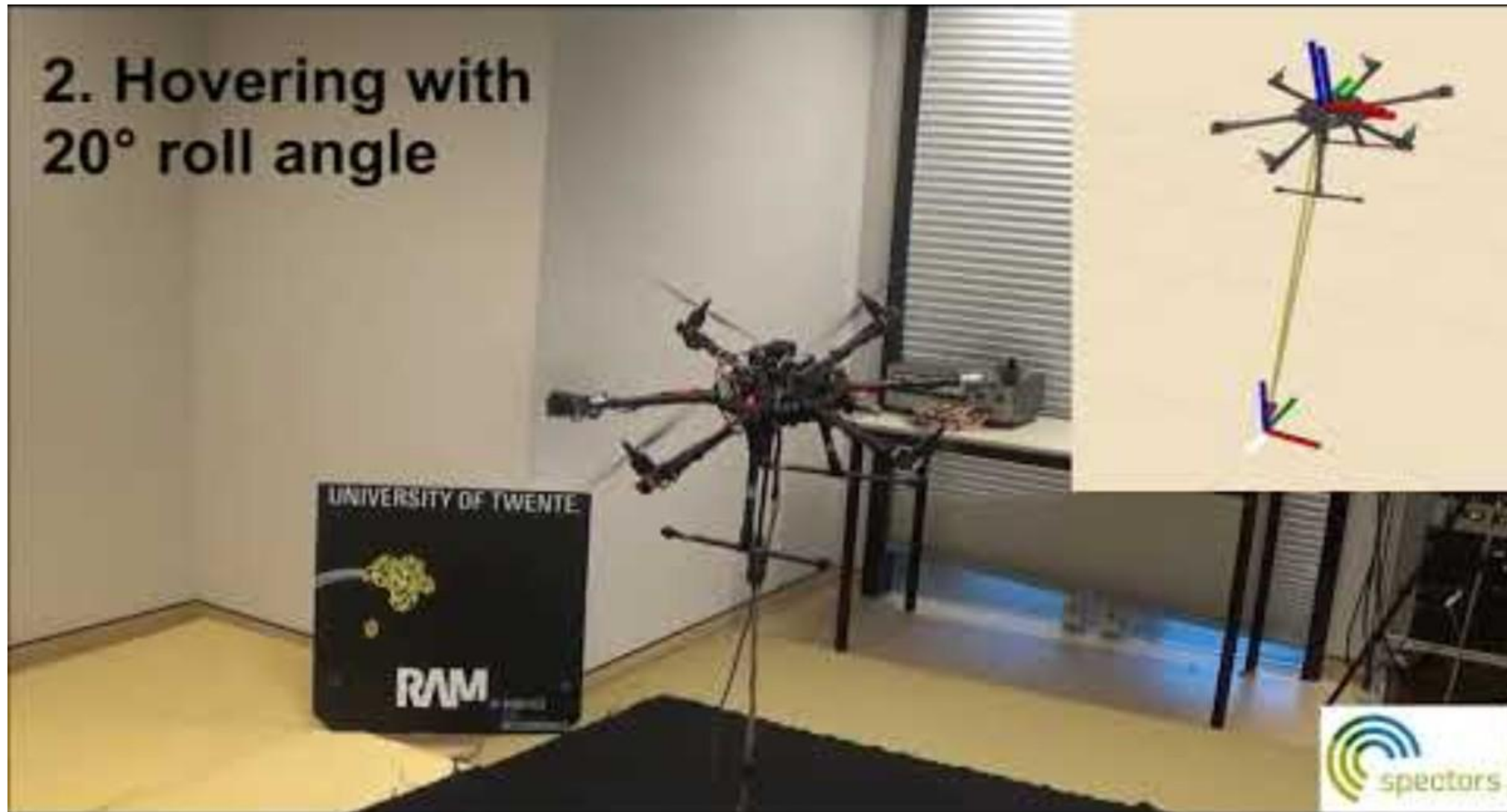
$$\mathcal{W}_{\text{con}}^{b,b} = \begin{pmatrix} \tau_{\text{con}}^{b,b} \\ f_{\text{con}}^{b,b} \end{pmatrix} = M(t)\lambda$$



$\lambda = (\lambda_1, \dots, \lambda_{N_p}) \in \mathbb{R}^{N_p}$ is the rotors thrust vector and N_p is the number of propellers on the MAV.



SE(3) Control of Fully-actuated MAV



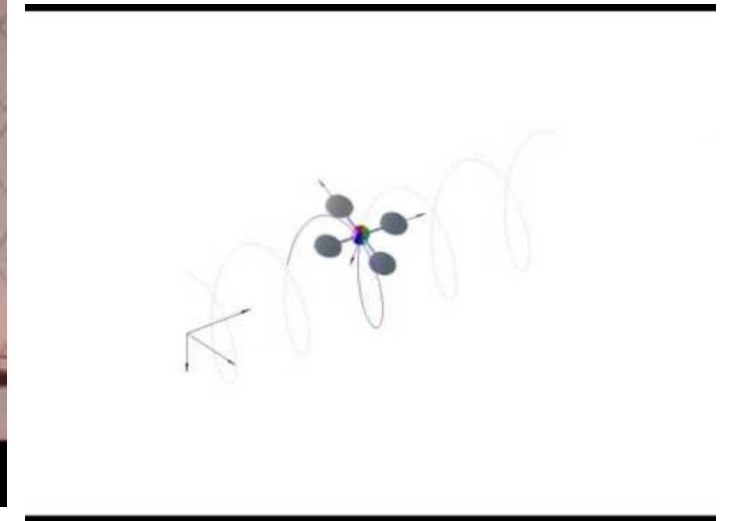
Rashad, R., Kuipers, P., Engelen, J., & Stramigioli, S. (2017). Design, modeling, and geometric control on SE (3) of a fully-actuated hexarotor for aerial interaction. *arXiv preprint arXiv:1709.05398*.



SE(3) Control of Under-actuated MAV



Mellinger, D., & Kumar, V. (2011, May). Minimum snap trajectory generation and control for quadrotors. In *2011 IEEE international conference on robotics and automation* (pp. 2520-2525). IEEE.



Lee, T., Leok, M., & McClamroch, N. H. (2010, December). Geometric tracking control of a quadrotor UAV on SE(3). In *49th IEEE conference on decision and control (CDC)* (pp. 5420-5425). IEEE.

